

# **Telemetry and Command system design for ST5: Approach, Lessons Learned thus Far**

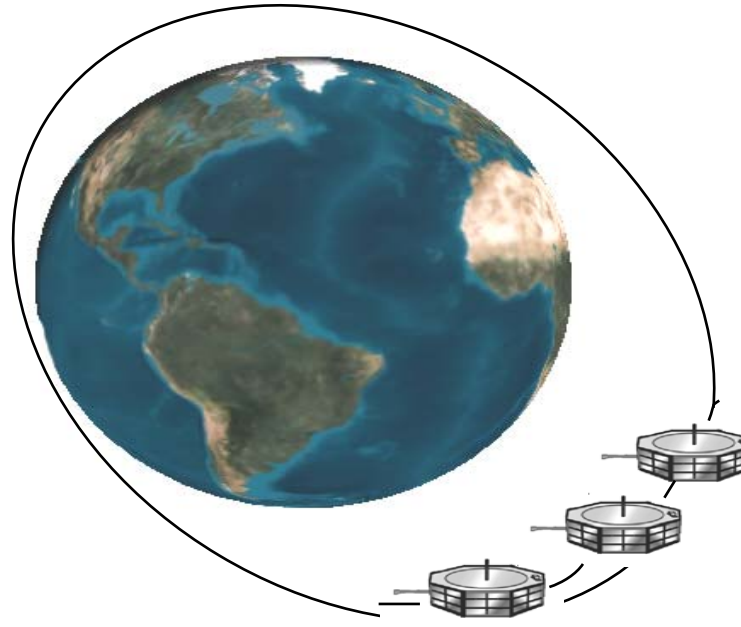
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# Agenda

- **The problem**
  - Requirements
  - Technical challenges
- **Our solution**
  - Design approach
  - Telemetry design
  - Command design
- **Lessons learned**
- **My role**
- **Summary**



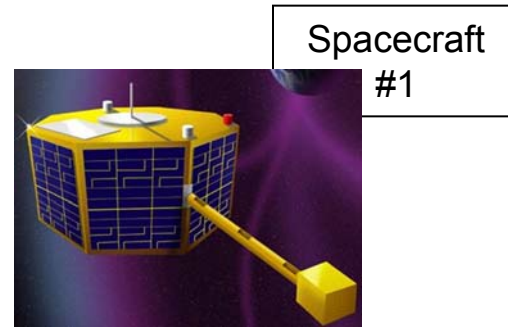
# ST5 New Baseline Orbit

- **Launch Timeframe:** February 28 - March 31, 2006
- **Launch Site:** Vandenberg AFB, Lompoc, CA
- **Mission Duration:** 90 days
- **Eclipses:** None due to earth shadow, March 29 eclipses on 2 - 3 orbits due to moon shadow
- **Perigee:** 300 km
- **Apogee:** 4500 km
- **Inclination:** 105.6 deg (sun synchronous)
- **Period:** 136 min
- **Number of orbits/day:** about 10.5
- **RAAN:** 68 for Mar 1 launch, increasing 1 deg/day for launch later in launch window (full sun 6 AM - 6 PM)
- **Argument of Perigee at Launch:** 160 deg
- **Rotation of Apsides:** -1.2 deg/day (into the southern hemisphere)
- **Constellation Configuration:** "String of Pearls"

# Requirements – ST5 Overview

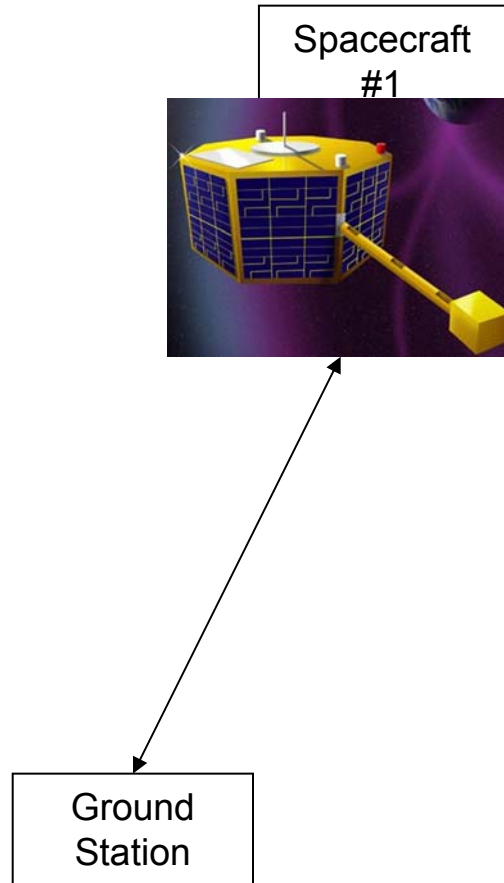
- 3 identical spacecraft
- Formation fliers
  - Only one pass at a time
  - Uplink/downlink with one spacecraft at a time
- CCSDS protocols
  - Unique Spacecraft-ID for each spacecraft
  - Spacecraft-ID is present in all frames (telemetry and command)
  - One telemetry bitstream per spacecraft (due to modulation scheme)

# ST5 Pass (multi-slide animation)

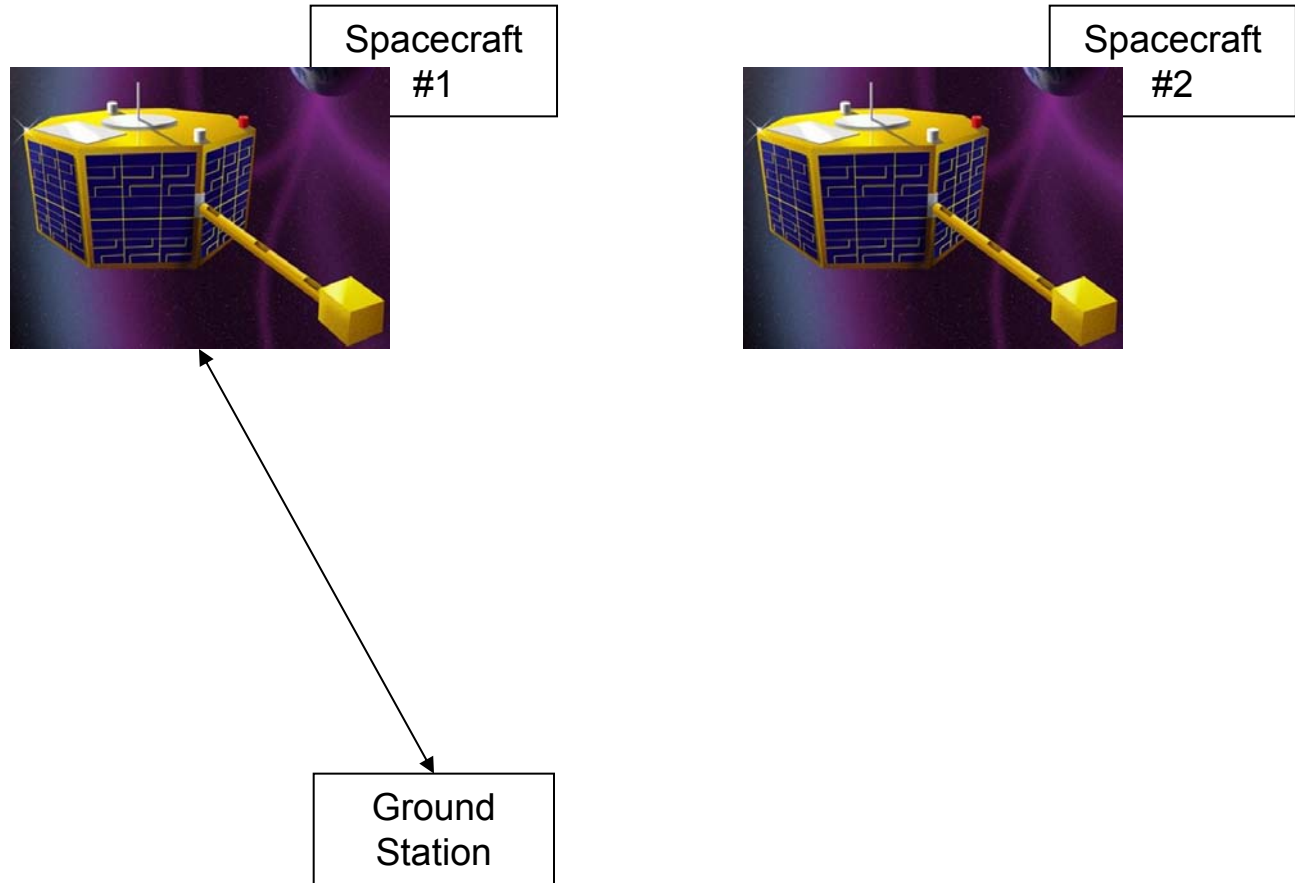


Ground  
Station

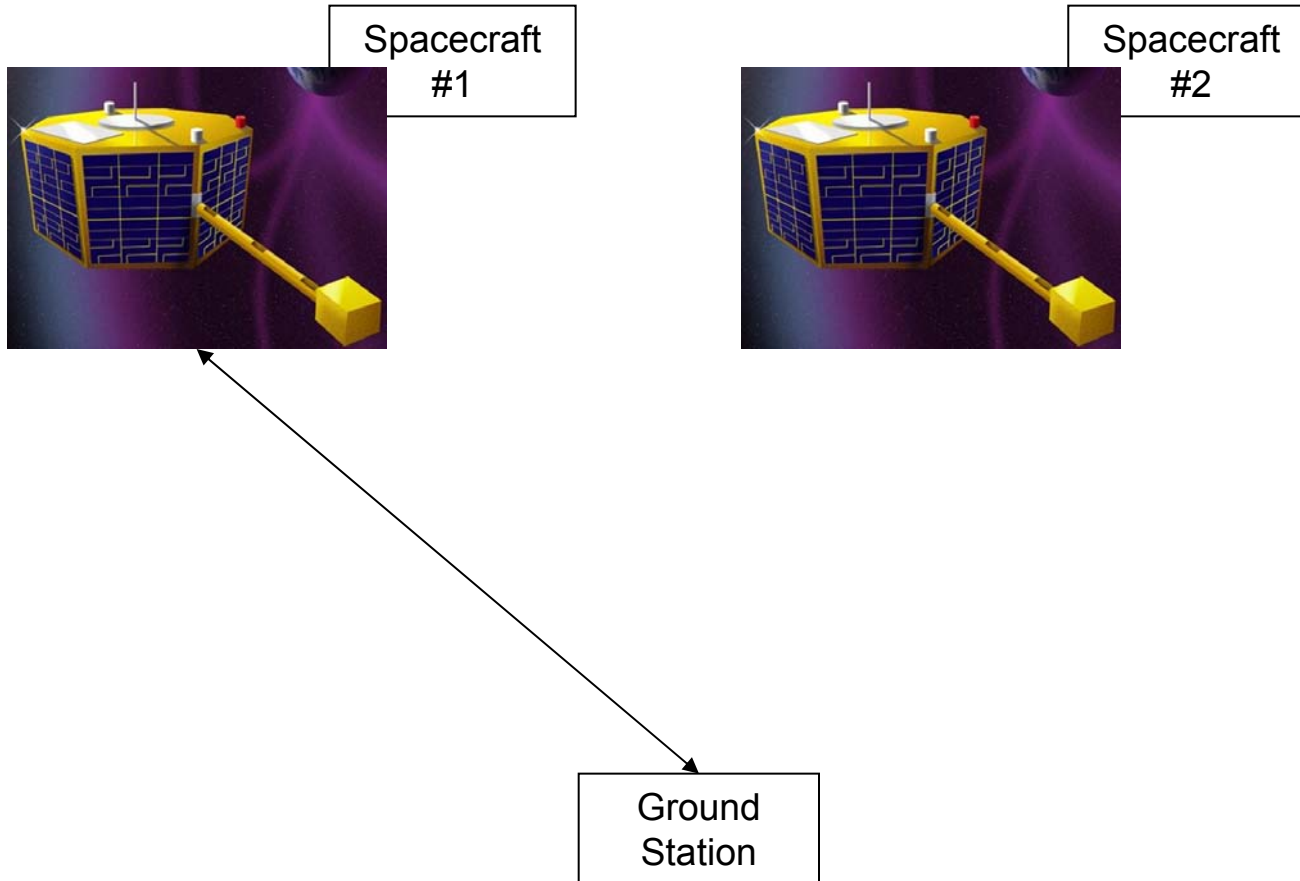
# ST5 Pass (multi-slide animation)



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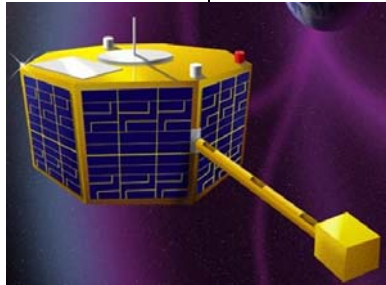
# ST5 Pass (multi-slide animation)



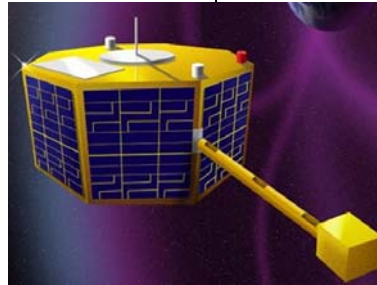


# ST5 Pass (multi-slide animation)

Spacecraft  
#1



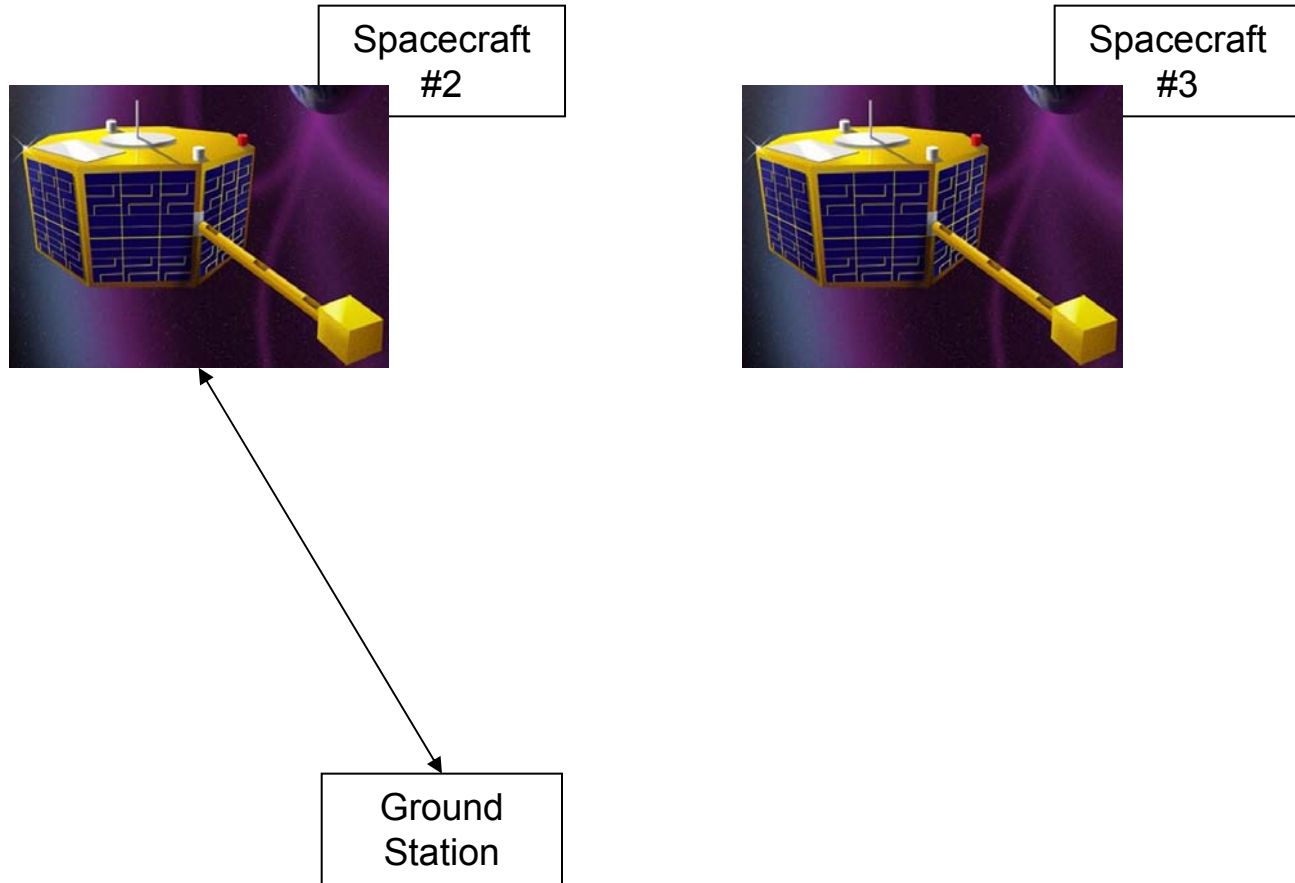
Spacecraft  
#2



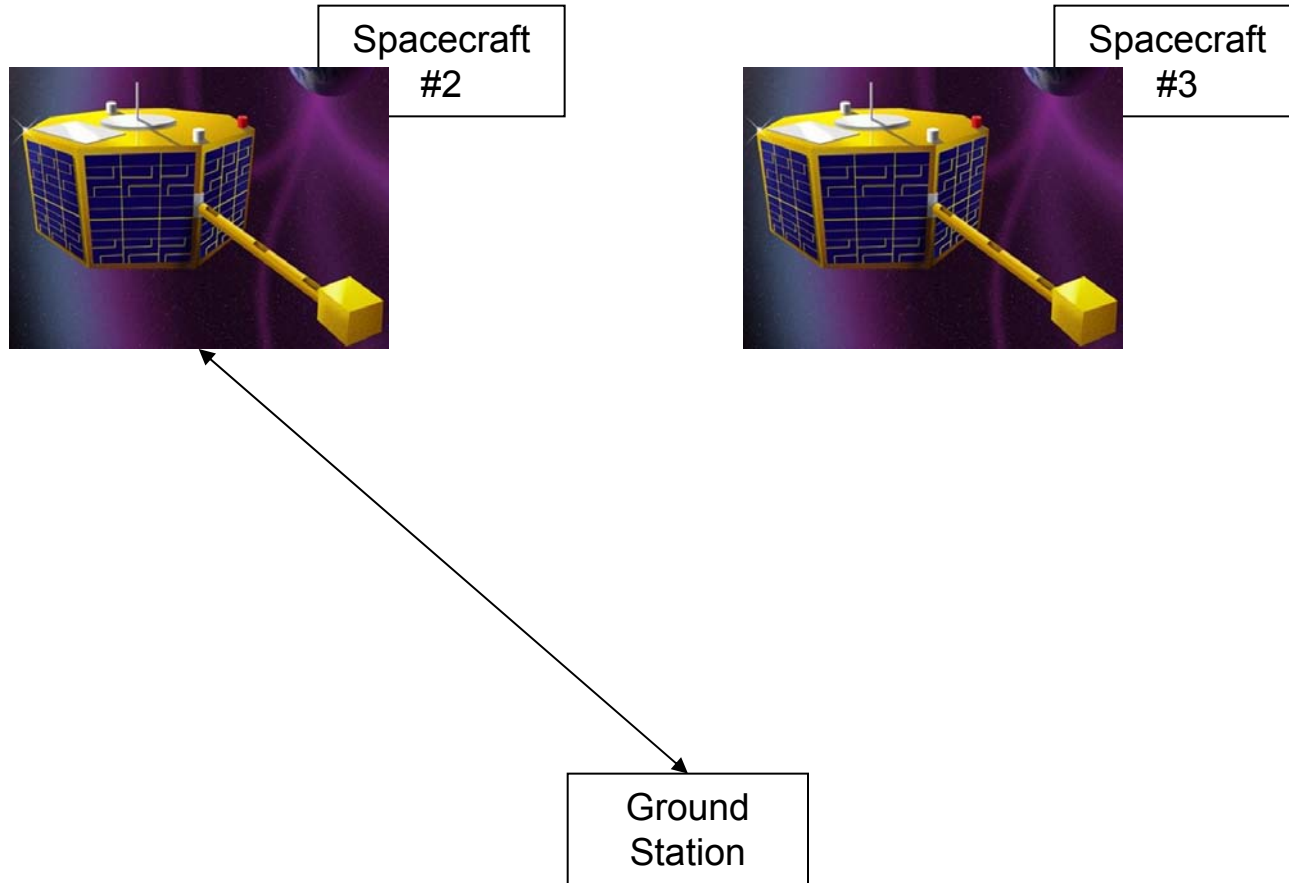
Ground  
Station



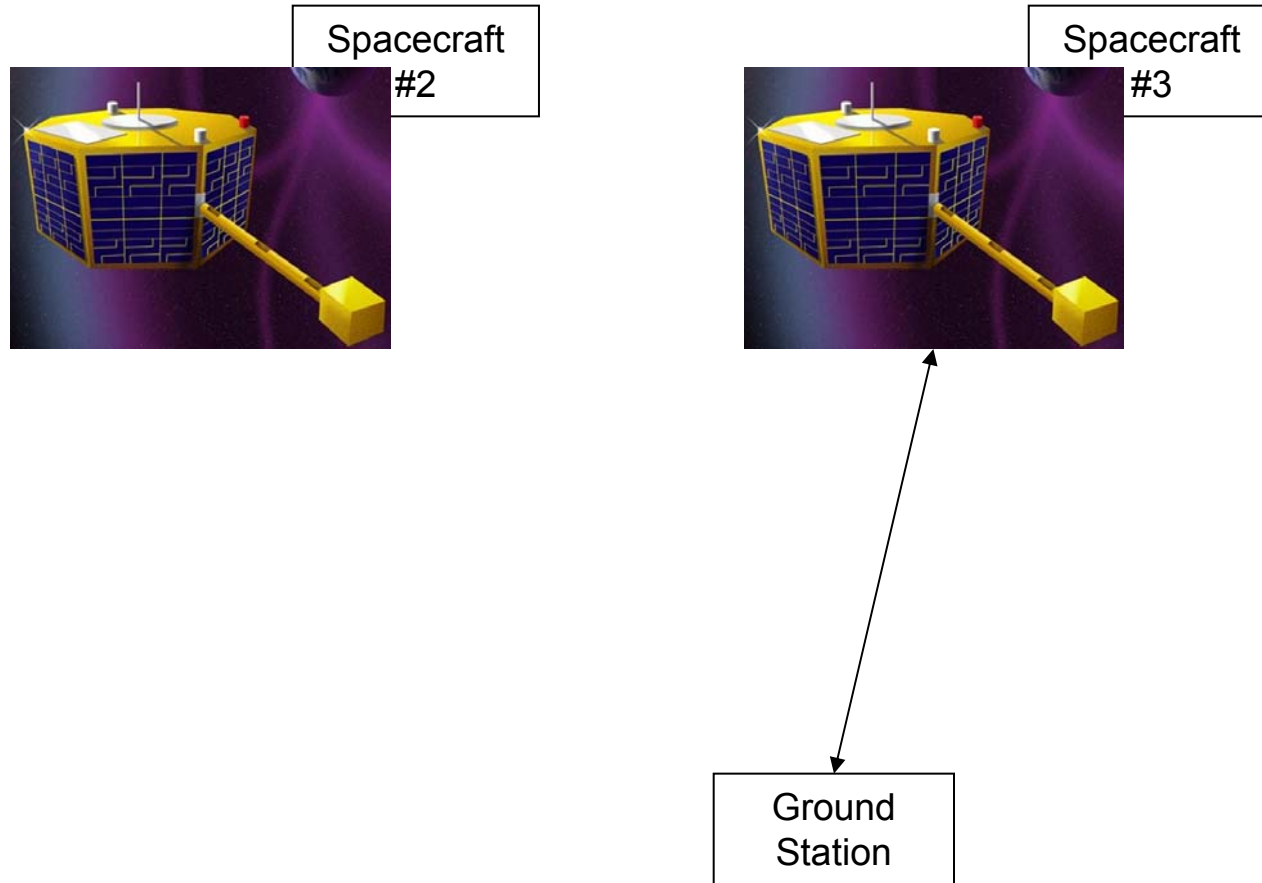
# ST5 Pass (multi-slide animation)



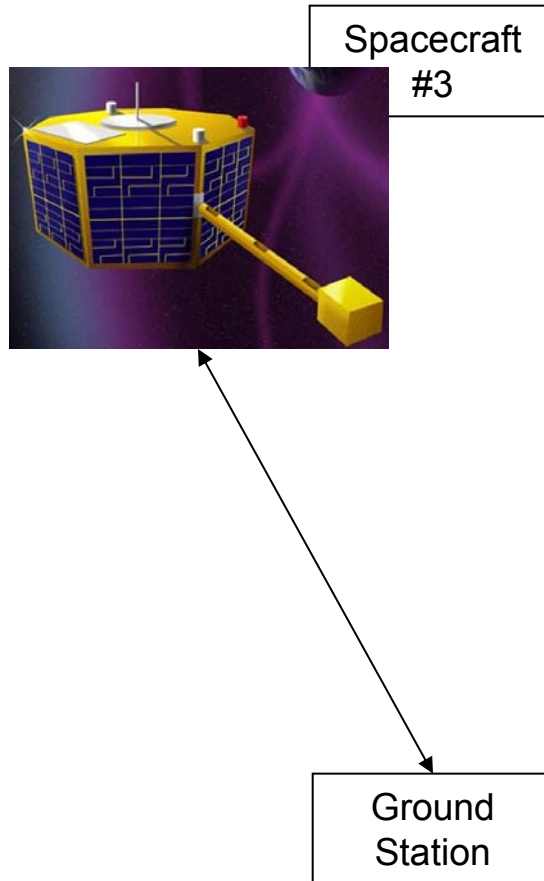
# ST5 Pass (multi-slide animation)



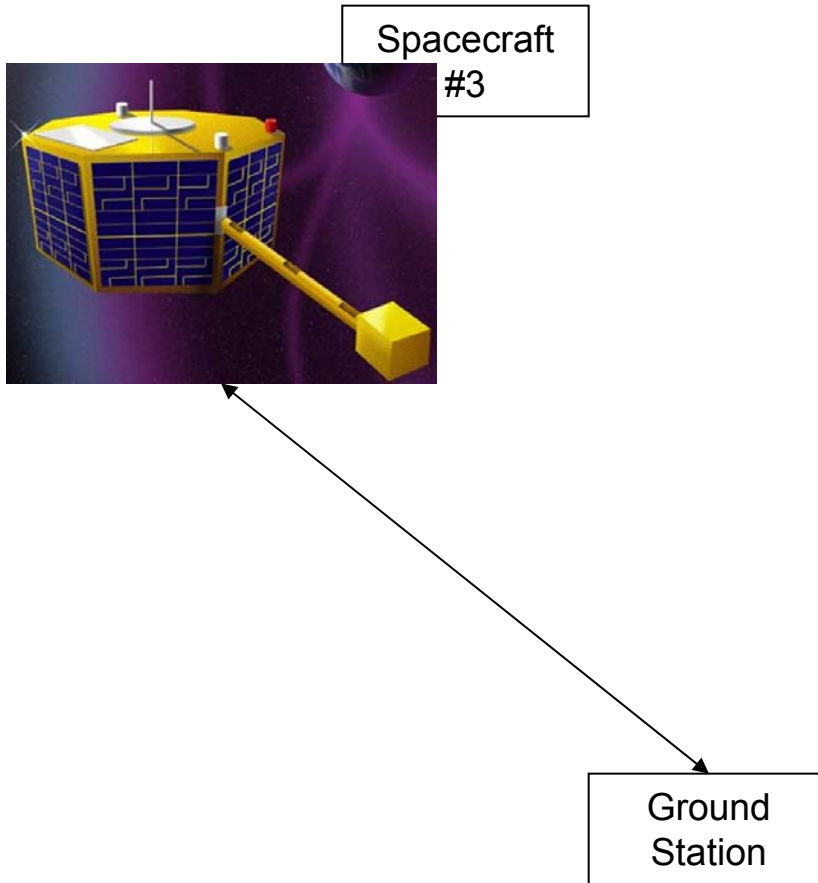
# ST5 Pass (multi-slide animation)



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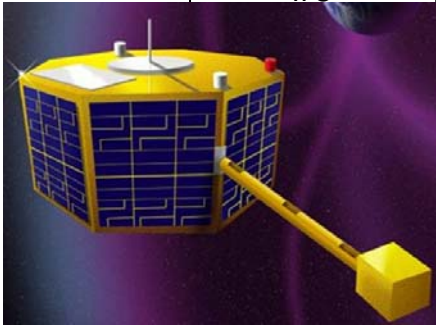


# ST5 Pass (multi-slide animation)



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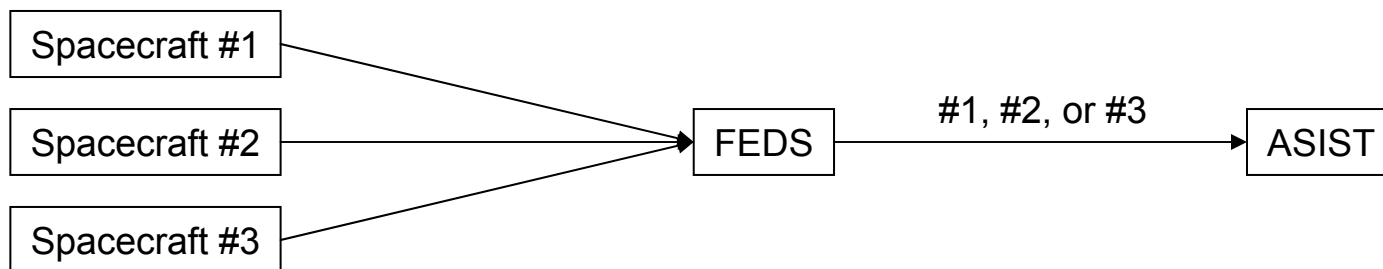
Spacecraft  
#3



Ground  
Station

## Requirements – ST5 telemetry

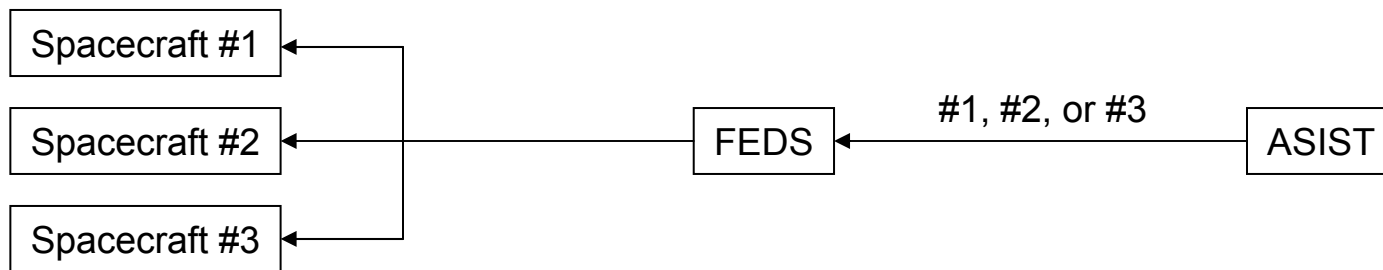
- I&T drives the requirements (parallel testing)
- Parallel ingest and processing (FEDS)
- Monitoring of telemetry values (ASIST)
  - Each ASIST workstation need only monitor one spacecraft at a time
  - User must be able to easily switch between spacecraft on-the-fly.
  - Parallel monitoring via multiple ASIST workstations





## Requirements – ST5 commanding

- Generate commands (ASIST)
  - Each ASIST workstation need only command one spacecraft at a time
  - User must be able to easily switch between spacecraft on-the-fly
  - Parallel commanding via multiple ASIST workstations
- Three concurrent COP-1 command feedback loops (FEDS)
- Single uplink (shared by all 3 spacecraft)



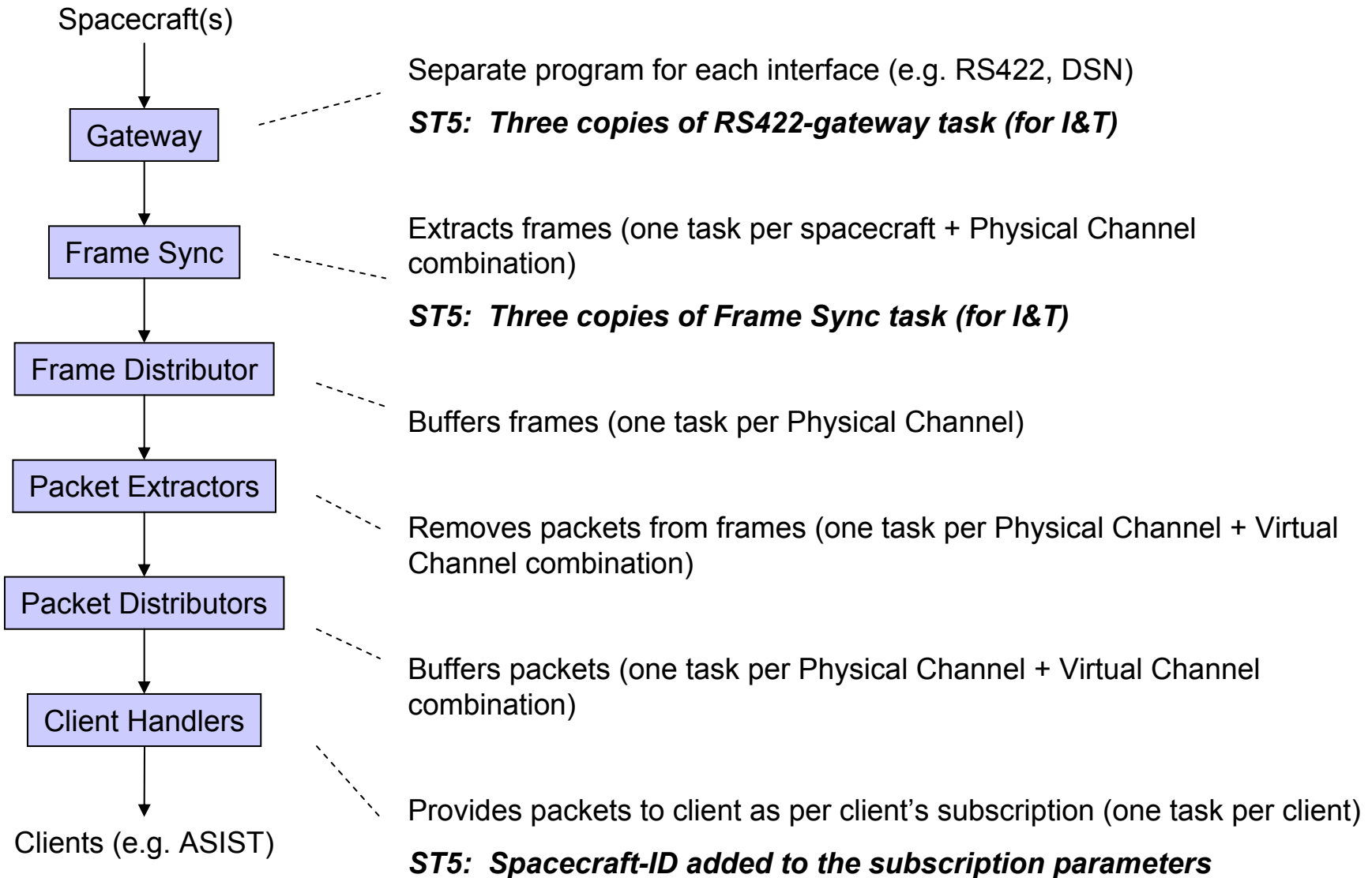
# Technical challenges

- I&T is more challenging than operations
  - Parallel versus serial “passes”
- Vast amount of existing capability; don’t break anything
  - Any change to existing source code may break existing capabilities.
- Maintaining backward compatibility (self-imposed requirements)
  - Product must handle both single-spacecraft and constellation missions ...
  - ... with identical source code.
  - New ASIST must work with old FEDS

# Design approach

- Generic solution
  - Handle a class of constellation missions (CCSDS spacecraft)
- Minimize changes to existing source code (rule of thumb)

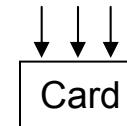
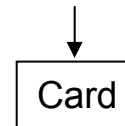
# Telemetry design (FEDS tasks)



# Telemetry design - Implementation

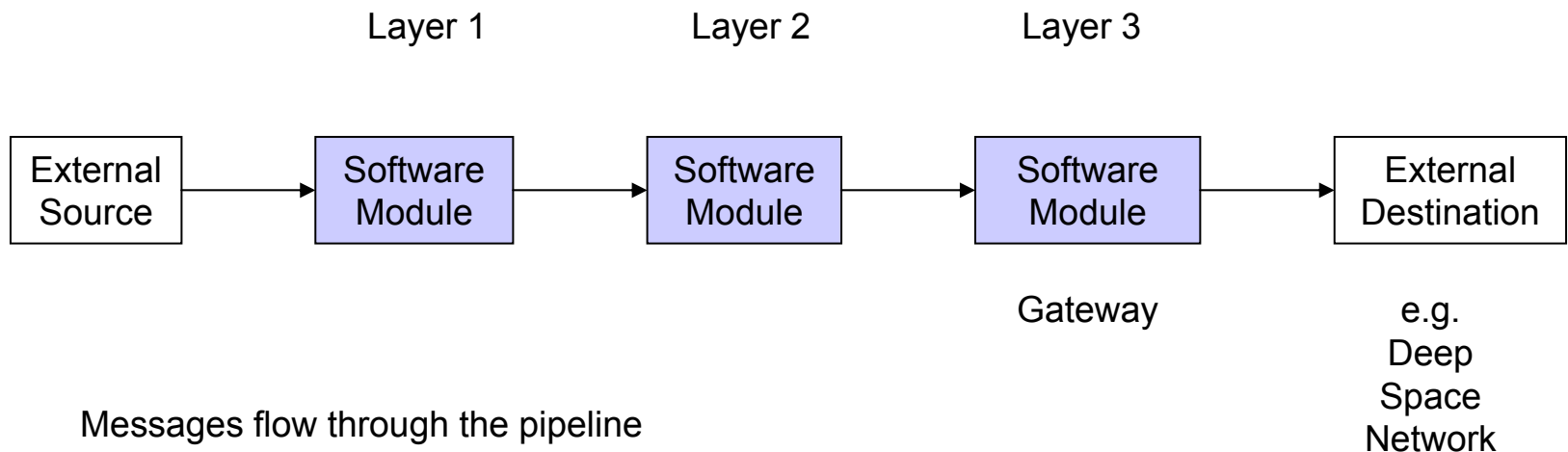
- Enhanced configuration capability
  - From hard-coded scripts to an XML-based utility

- Enhanced RS422 driver
  - From one input to multiple concurrent inputs



- Added Spacecraft-ID to various algorithms/interfaces
- Enhanced command feedback
  - From one path to multiple paths

# Commanding – Pipeline Architecture concept



Messages flow through the pipeline

Each module performs work (may include message conversion)

Message format is always the same...

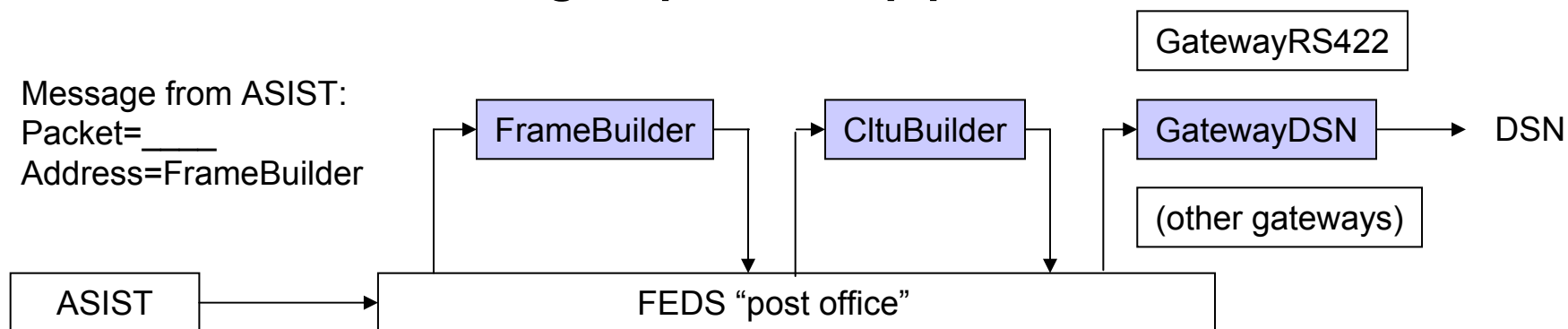
... except for Gateway module (e.g. match Deep Space Network interface)

There can be multiple modules per layer...

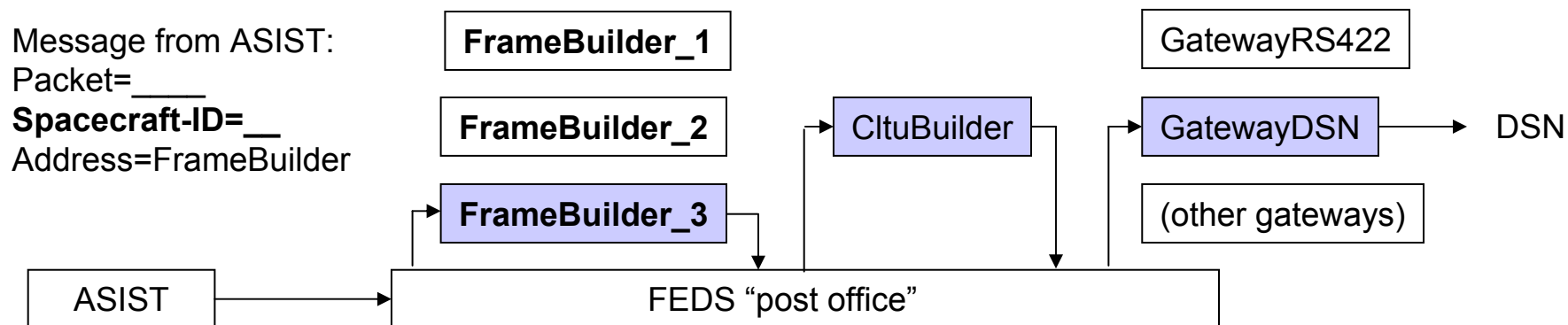
... but each message goes through one module per layer.

# Command design – Pipeline Architecture

## Single-spacecraft pipeline



## Constellation (ST5) pipeline



CLTU = Command Link Transmission Unit (bitstream)

# Command design - Implementation

- Added Spacecraft-ID field to ASIST command message format
- Added address-translation (routing) capability to FEDS “post office”
  - Generic capability; specifics via configuration files
  - Can support routing to one of n concurrent Ground Station passes
- Enhanced command feedback
  - From one path to multiple concurrent paths
- Very few source code changes
  - Less than one person-month each for ASIST and FEDS



# Lessons Learned

- Design with an eye towards possible future enhancements
- Provide generic capabilities
  - Our Pipeline Architecture enabled an easy transition from commanding single-spacecraft to commanding a constellation

# My role

- Co-designer of the Pipeline Architecture (with Jeff Condron)
- Implementer of all the FEDS command software

## Summary -- Success

- New ASIST/FEDS delivered on-schedule to ST5 I&T
  - I&T: June 2004 – January 2006
  - Launch: Spring 2006
- Backward-compatible to single-spacecraft missions
  - Delivered to SDO labs (using identical source code)
- Supports an entire class of constellations (of CCSDS spacecraft)
  - With identical source code
  - Mission-specifics via configuration files